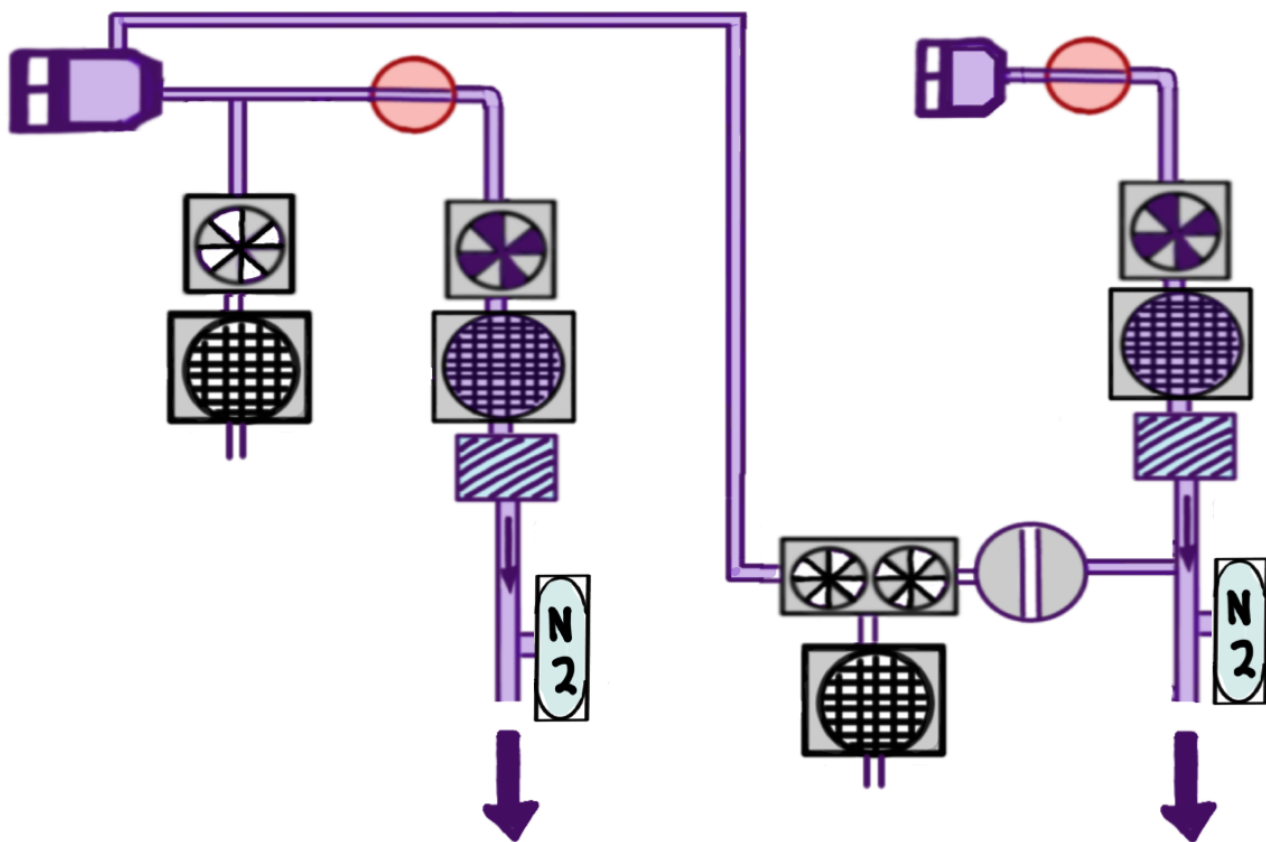


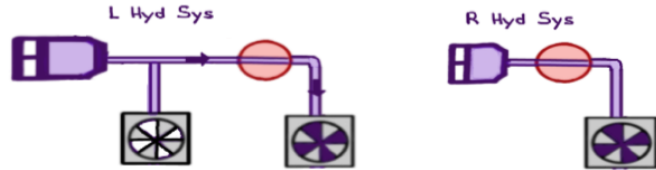
G500 HYDRAULIC SYSTEM



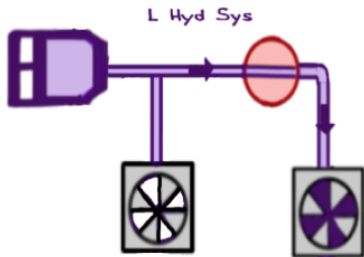
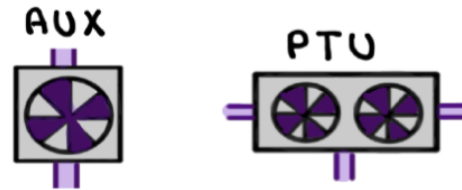
For study purposes only

The Hydraulic System is about the STORAGE AND DELIVERY of hydraulic fluid (Skydrol) UNDER high PRESSURE TO ACTUATE VARIOUS SYSTEMS

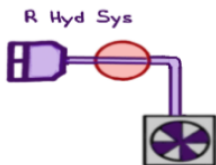
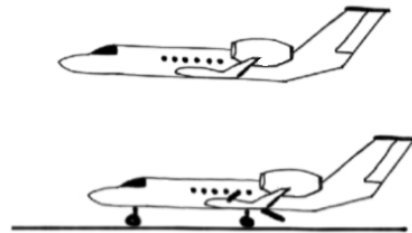
Two (2) MAIN SYSTEMS



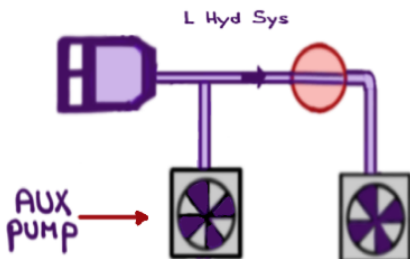
SUPPORTED by Two (2) sub-systems



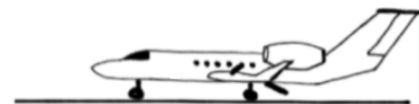
FLY
LAND



FLY



LAND
MX

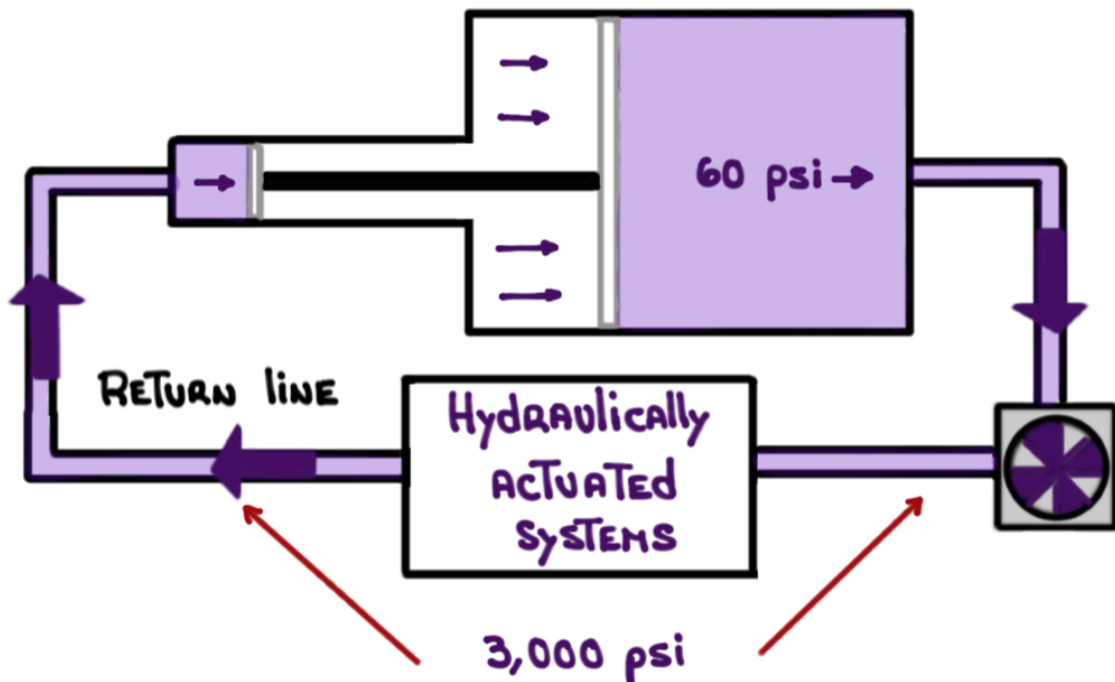


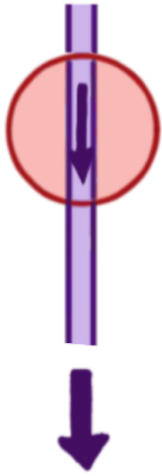
HYDRAULIC SYSTEM COMPONENTS



RESERVOIR: To store fluid

- COMPRESSED by bootstrap to PREVENT hydraulic pump CAVITATION
- LOCATED in THE TAIL COMPARTMENT
- SYSTEM MUST BE PRESSURIZED for ACCURATE quantity checks
- SINGLE CHAMBER





Shutoff valve: To shutoff hydraulic fluid To The ENGINE IN THE EVENT of ENGINE FIRE OR failure

- LOCATED IN THE TAIL COMPARTMENT AND ACTIVATED VIA FIRE HANDLES



Pump: To PRESSURIZE SYSTEM

- ENGINE-DRIVEN PUMP
- LOCATED IN THE ENGINE'S GEARBOX
- 3,025 ± 50 Psi



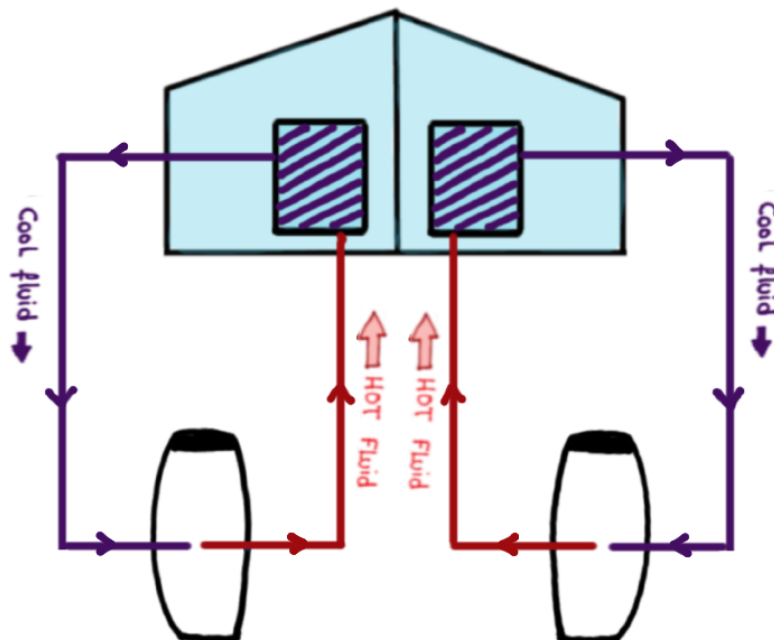
Filter Manifold: To filter hydraulic fluid AND CONTROL direction of flow

- LOCATED IN THE TAIL COMPARTMENT
- LEFT Hydraulic SYSTEM: Six (6) filters
- Right Hydraulic SYSTEM: THREE (3) filters

Hydraulic fluid-To-fuel HEAT EXCHANGER:

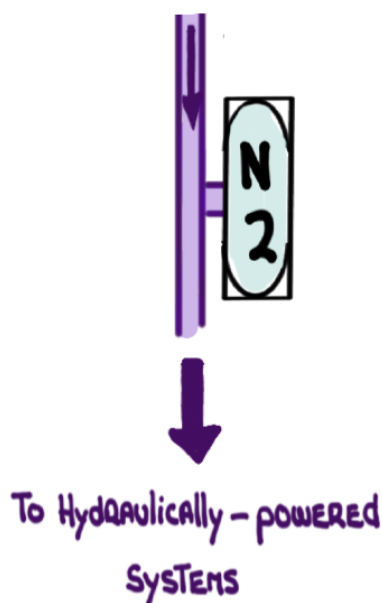
To cool hydraulic fluid AND To WARM UP cold fuel

- LOCATED IN THE ON-SIDE FUEL Hopper
- CONTINUOUS flow



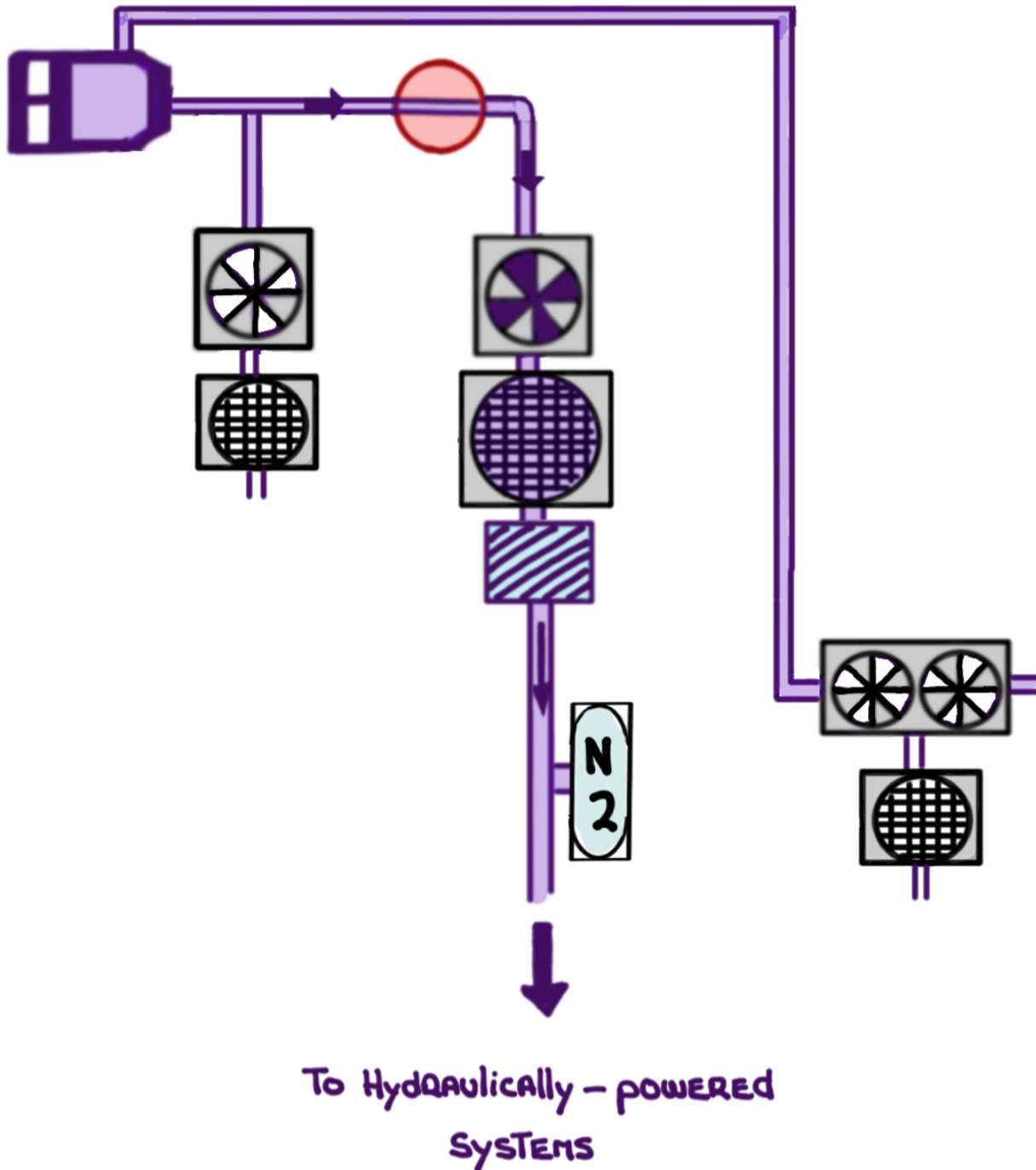
Accumulator: To absorb system shocks

- PRE-CHARGED To 1,200 Psi @ 70°F
- Absorbs fluid shocks within THE system
- SERVICED with Nitrogen
- LOCATED IN THE TAIL COMPARTMENT



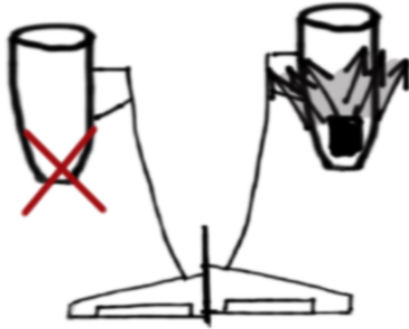
L HYDRAULIC SYSTEM

- INDEPENDENT AND ISOLATED FROM RIGHT HYDRAULIC SYSTEM
- SUPPORTED BY THE AUX PUMP AND PTU SUB-SYSTEMS



- POWERED by THE LEFT ENGINE-driven pump (EDP)

- MOUNTED ON ENGINE GEARBOX
- CONSTANT PRESSURE, VARIABLE VOLUME PUMP
- PRESSURIZES fluid To $3,025 \pm 50$ Psi
- FLOW RATE VARIES BASED ON POWER SETTING
- FAILURE of EDP RESULTS in loss of:



① LEFT THRUST REVERSER




② Mid spoiler panels

(285 KCAS / $M0.90$ MAXIMUM)

- Offload FEATURE:

- AUTOMATICALLY CONTROLLED BY ELECTRONIC ENGINE CONTROLLER (ECC)
- REDUCES PUMP OUTLET PRESSURE IN flight when ENGINE DROPS below idle (< 55% N₂)
- REDUCES DRAG ON ENGINE TO MAXIMIZE AIRSTART capability
- No windmilling PRESSURE

- LARGEST RESERVOIR:

- CONSIDERED full AT 2.4 gallons 
- MUST BE PRESSURIZED FOR ACCURATE READING
- SUPPLIES HYDRAULIC fluid To:

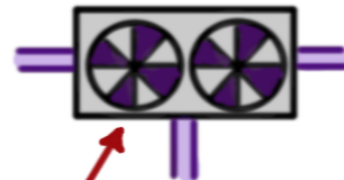
LEFT EDP



AUX
PUMP



PTU

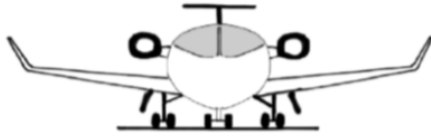


Pump



- Majority of AIRCRAFT Hydraulic functions:

• LANDING GEAR



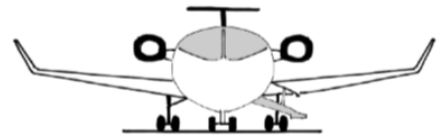
• BRAKES



• FLAPS



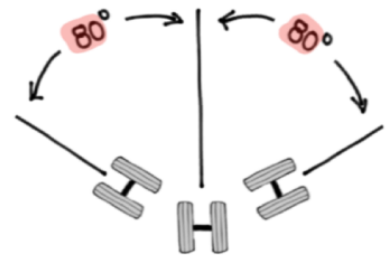
• MAIN DOOR



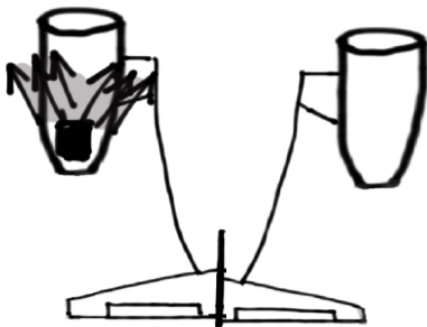
• MID SPOILER PANELS



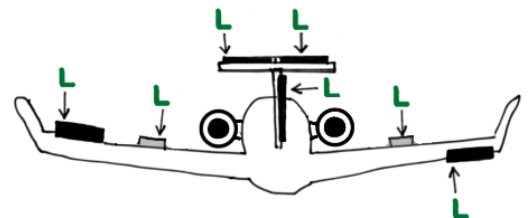
• NOSE WHEEL STEERING



• LEFT THRUST REVERSE

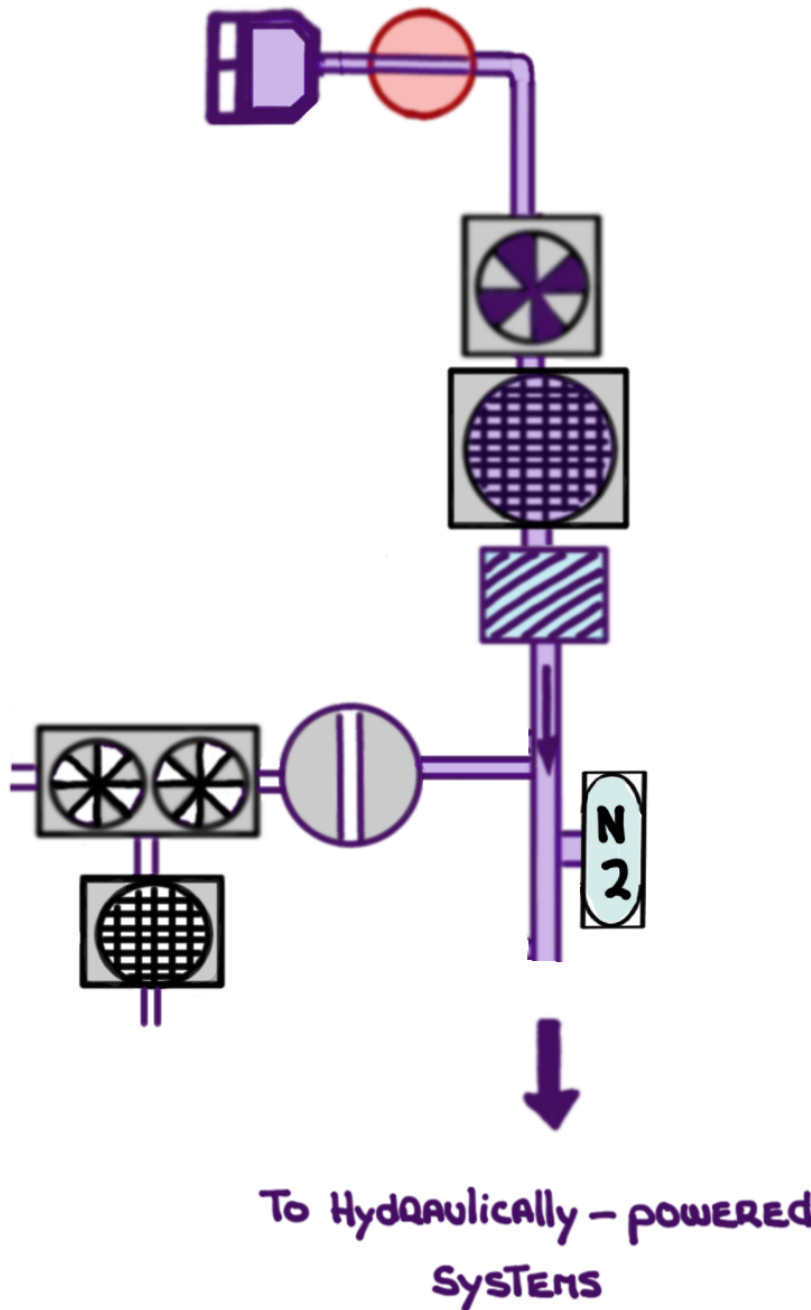


• FLIGHT CONTROLS



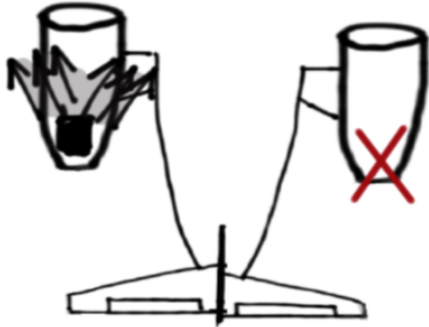
R Hydraulic System

- Independent and isolated from LEFT Hydraulic System



- POWERED by the Right ENGINE - driven pump (EDP) (EDP)

- MOUNTED ON ENGINE GEARBOX
- CONSTANT PRESSURE, VARIABLE VOLUME PUMP
- PRESSURIZES fluid TO **3,025 ± 50** PSI
- FLOW RATE VARIES BASED ON POWER SETTING
- FAILURE OF EDP RESULTS IN LOSS OF:



① RIGHT THRUST REVERSER




② INBOARD SPOILER PANELS

(**285** KCAS / **M0.90** MAXIMUM)

- Offload FEATURE:

- AUTOMATICALLY CONTROLLED BY ELECTRONIC ENGINE CONTROLLER (ECC)
- REDUCES PUMP OUTLET PRESSURE IN flight when ENGINE DROPS BELOW idle (< 55% N₂)
- REDUCES DRAG ON ENGINE TO MAXIMIZE AIRSTART CAPABILITY
- No windmilling PRESSURE

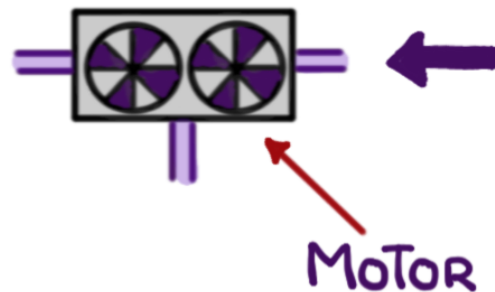
- SMALLEST RESERVOIR:

- CONSIDERED full AT 1.3 gallons 
- MUST BE PRESSURIZED FOR ACCURATE READING
- SUPPLIES HYDRAULIC FLUID TO:

Right EDP



PTU

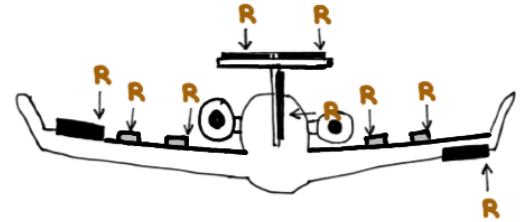


- ACTUATES THE following AIRCRAFT Hydraulic functions:

• BRAKES



• FLIGHT CONTROLS



• INBOARD/OUTBOARD PANELS

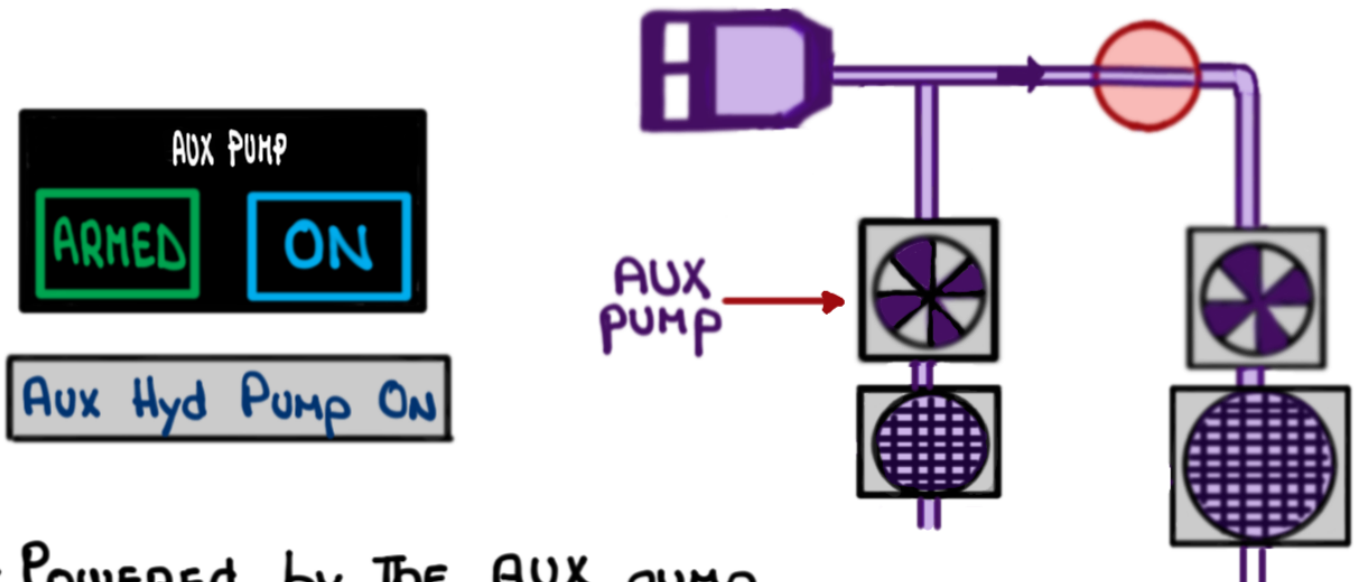


• RIGHT THRUST REVERSE



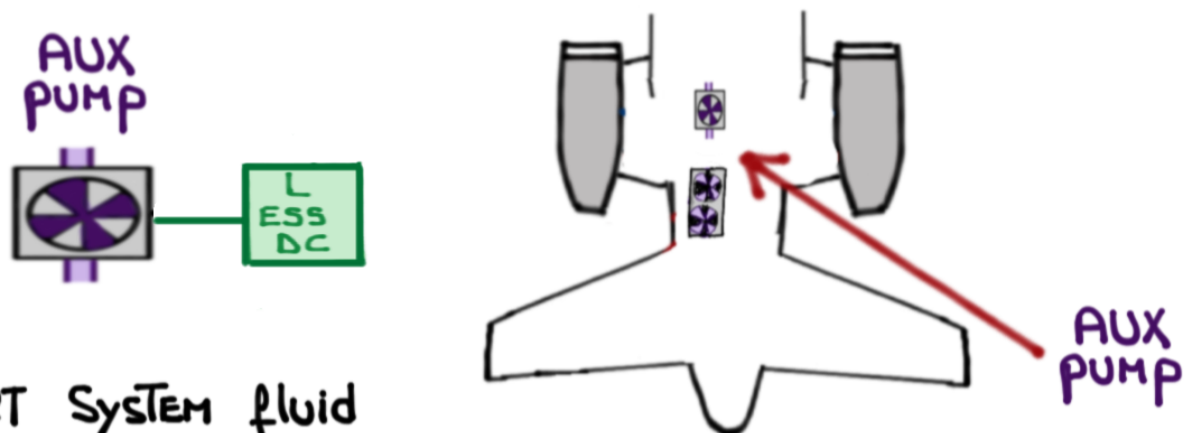
AUXILIARY (AUX) HYDRAULIC SYSTEM

- SUPPLEMENTS THE LEFT HYDRAULIC SYSTEM



- POWERED BY THE AUX PUMP

- LOCATED IN THE TAIL COMPARTMENT BELOW THE LEFT HYDRAULIC RESERVOIR
- ELECTRICALLY POWERED BY LESS DC BUS



- USES LEFT SYSTEM FLUID

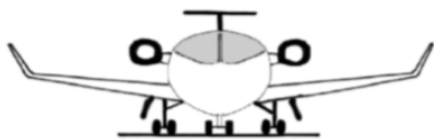
- OPERATES EITHER AUTOMATICALLY OR MANUALLY

- **3,000** PSI @ TWO POINT FIVE (**2.5**) GALLONS PER MINUTE

- PRIMARY function:

Hydraulic pressure for utility systems during **GROUND** AND MAINTENANCE ACTIVITIES

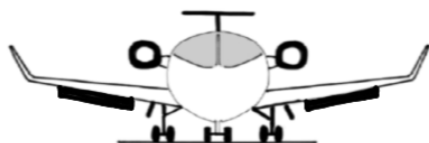
• LANDING GEAR



• BRAKES



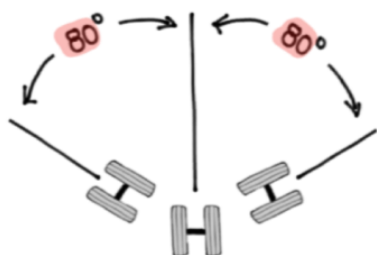
• FLAPS



• MAIN DOOR



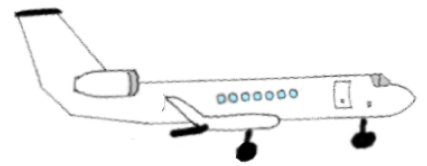
• NOSE WHEEL STEERING



- SECONDARY function:

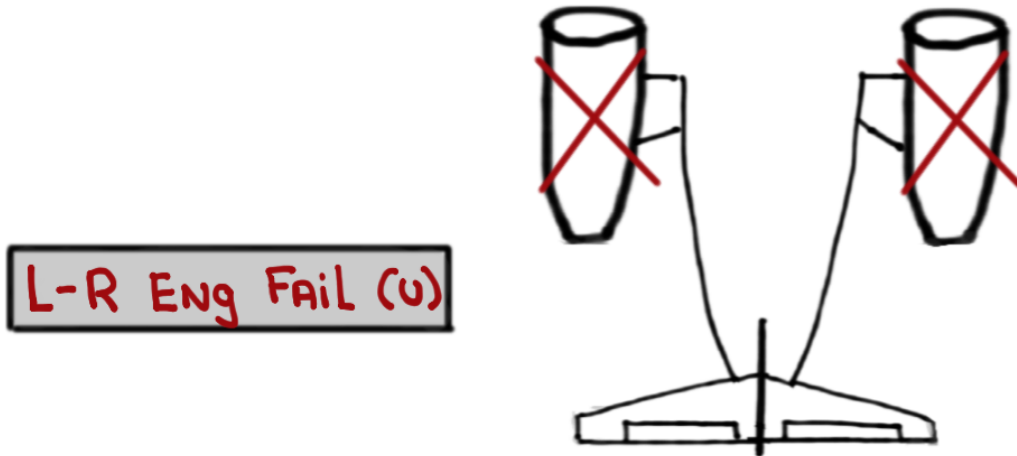
In flight back up to assist the  if NECESSARY

- AUTO OPERATION IN FLIGHT:

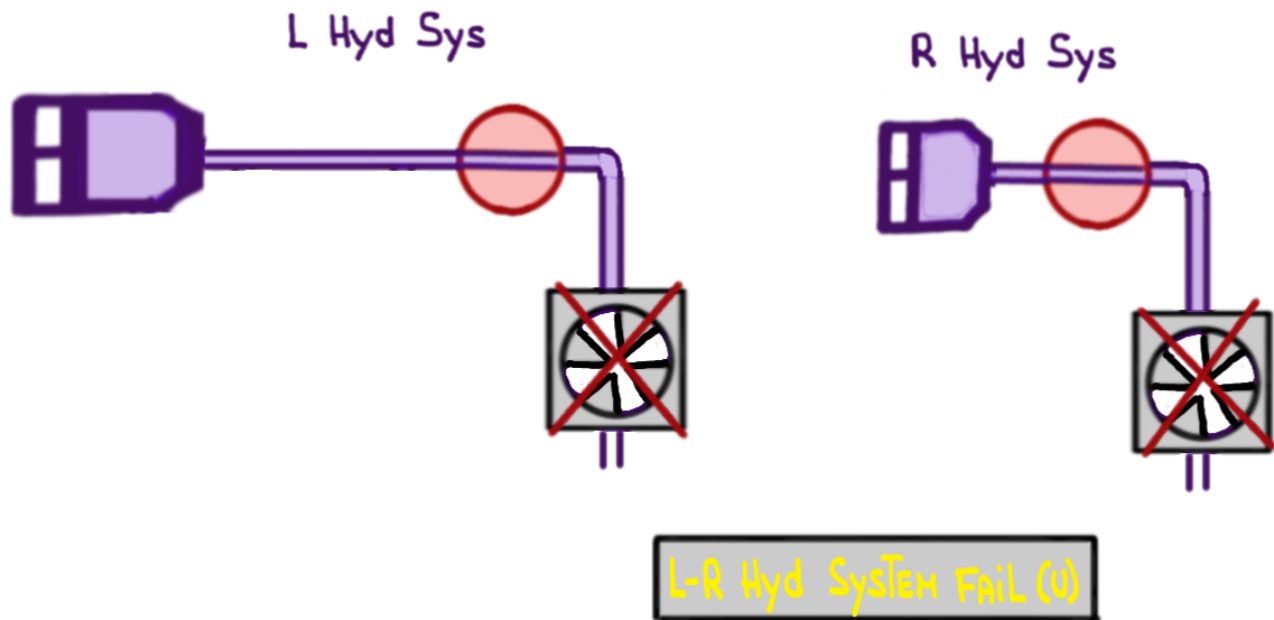


NORMALLY INACTIVE in flight BUT will power **ON** AUTOMATICALLY FOR THE OPERATION OF LANDING GEAR AND flaps

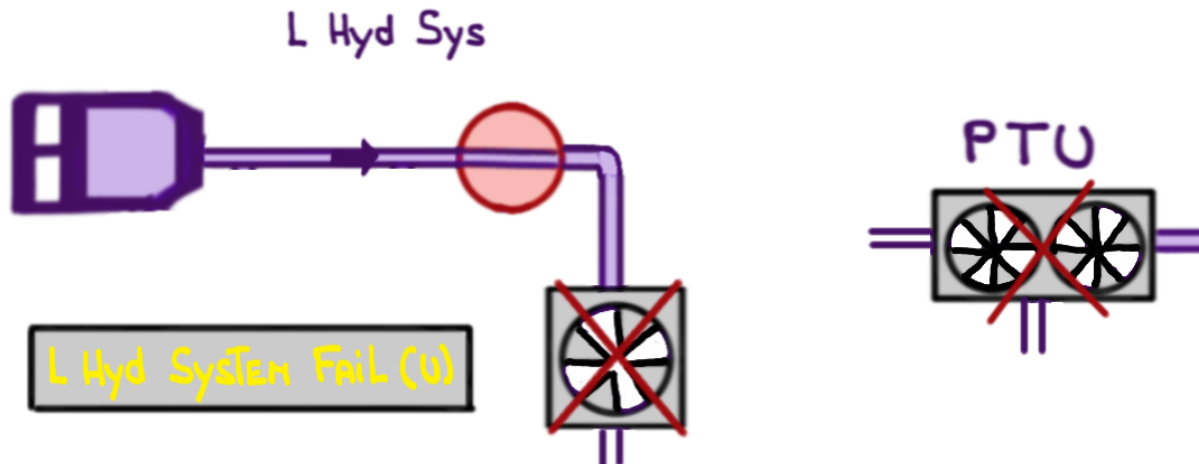
① DUAL ENGINE FAILURE



② DUAL ENGINE-DRIVEN PUMP FAILURE



③ LEFT ENGINE-DRIVEN PUMP AND PTU FAILURE



- REQUIREMENTS FOR AUTO ON OPERATION:

- AUX PUMP ARMED (DEFAULT MODE AT POWER UP)
- AUX PUMP NOT OVERLOADED / OVERHEATED
- LEFT SYSTEM PRESSURE < 1,500 Psi
- LEFT SYSTEM fluid AVAILABLE AND NOT HOT
(> 0.36 gallons AND < 107°C)
- Flaps OR GEAR POSITION DOES NOT MATCH HANDLE
POSITION > 100 KCAS

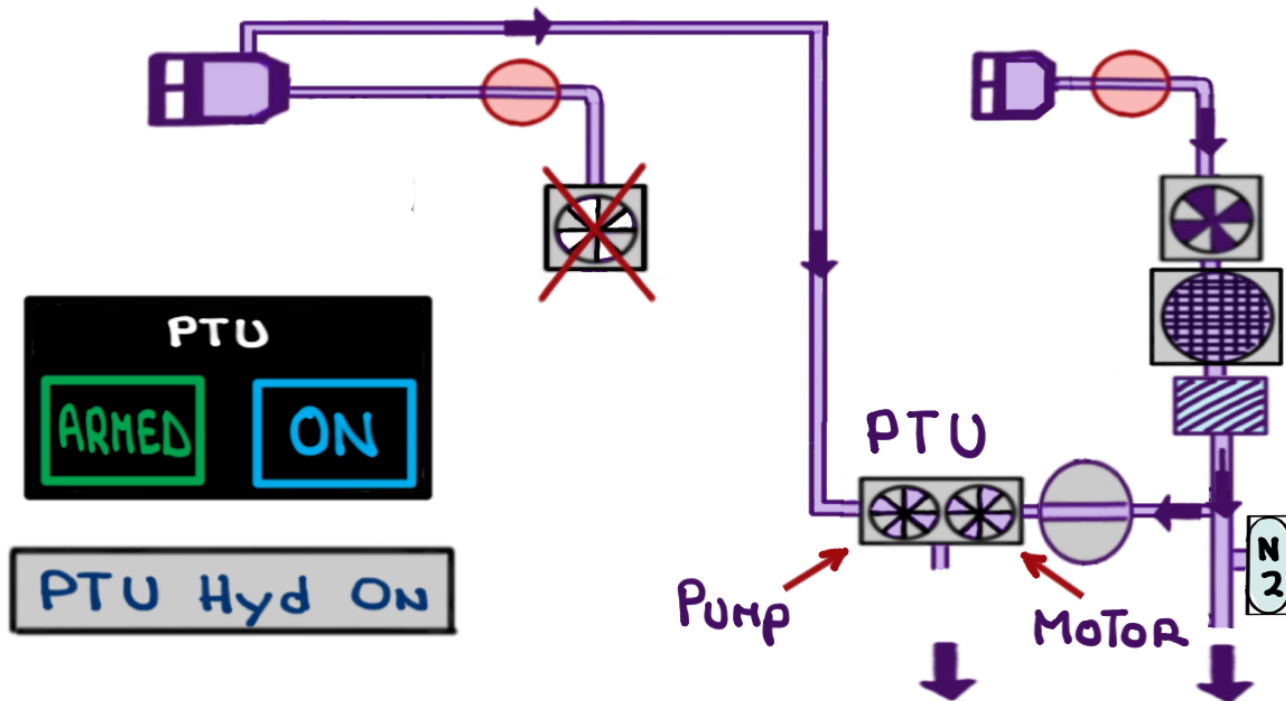
- AFTER THE flaps OR GEAR REACHES ITS SELECTED POSITION
THE AUX PUMP SWITCHES ITSELF **OFF**

- OPERATION LIMITATIONS:

- In flight when the AUX pump has been manually selected ON it will go OFF AFTER TWO (2) MINUTES of operation. THE TIMER CAN BE RESET by TURNING THE AUX pump OFF THEN ON
- THERE ARE NO TIME LIMITATIONS ON THE GROUND

POWER TRANSFER UNIT (PTU)

- Back up To The LEFT Hydraulic System engine - driven pump (OPERATIONAL REDUNDANCY)



- The **PTU**  is a MOTOR/pump ASSEMBLY

The MOTOR is DRIVEN by Right System PRESSURIZED fluid. The pump is DRIVEN by THE MOTOR AND its job is TO PRESSURIZE LEFT System fluid

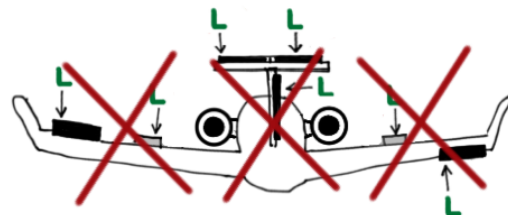
- IT COMES ON AUTOMATICALLY if L Hydraulic System PRESSURE is < 2,400 PSI

- IT CANNOT ACTUATE:

LEFT THRUST REVERSE



FLIGHT CONTROLS AND MID SPOILERS



PTU

- THE  CANNOT OPERATE WITHOUT:

- ▶ **L** HYDRAULIC SYSTEM FLUID
- ▶ **R** HYDRAULIC SYSTEM FLUID AND PRESSURE

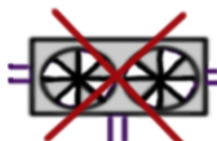
PTU

- THE  IS PREVENTED FROM COMING ON AUTOMATICALLY IF:

L Hyd Sys



< 0.36 g



R Hyd Sys



> 107°C

< 2,850 psi

- OPERATES EITHER AUTOMATICALLY OR MANUALLY
- **3,000** Psi @ **23** gallons PER MINUTE
- HELPS RETRACT THE LANDING GEAR FOLLOWING A FAILURE OF THE LEFT ENGINE AFTER TAKEOFF (REGULATORY REQUIREMENT)

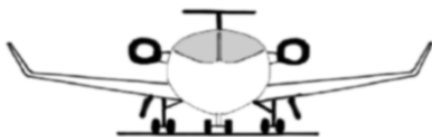
PTU = "Pick TIRES Up"

- IT USES:

- ① RIGHT SYSTEM PRESSURIZED fluid, AND
- ② LEFT SYSTEM fluid

- IT CAN ACTUATE:

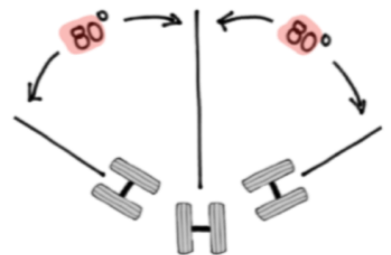
LANDING GEAR



FLAPS



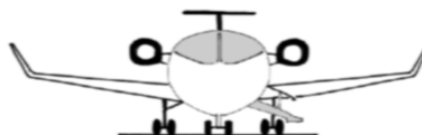
NOSE WHEEL STEERING

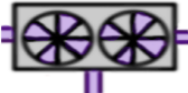


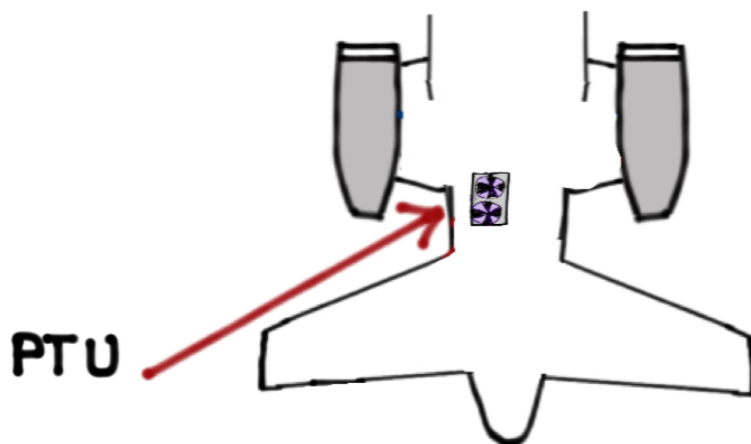
BRAKES



MAIN DOOR



- The **PTU**  is located in the TAIL COMPARTMENT



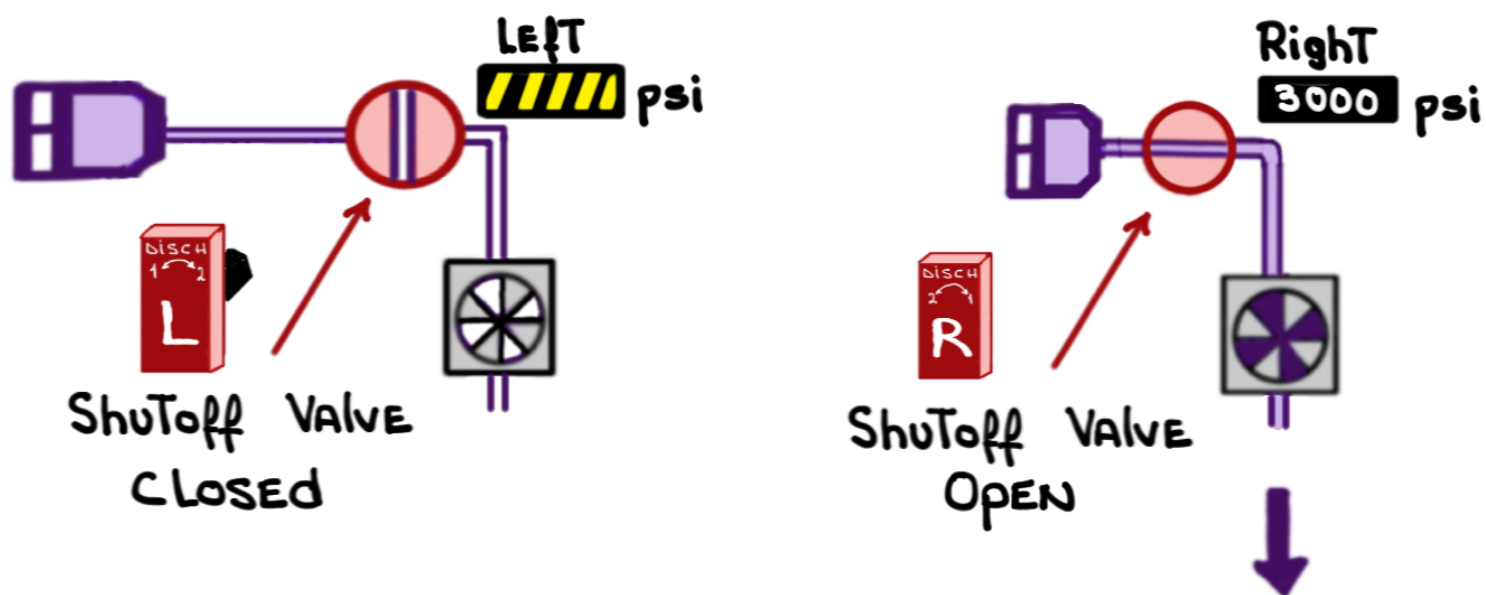
- When ARMED it has a SEVEN (7) SECOND DEBOUNCE. This MEANS THAT IT WILL RUN AT LEAST SEVEN (7) SECONDS TO PREVENT INTERMITTENT OPERATION WITH FLUCTUATING LEFT SYSTEM PRESSURE. This is CONTROLLED BY THE DCN

DEACTIVATION:

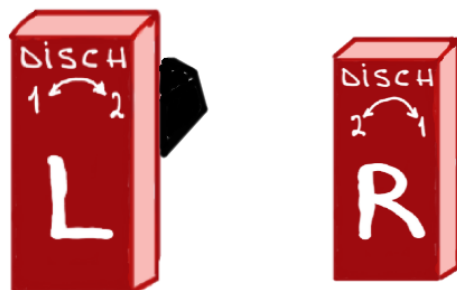
- ▶ SEVEN (7) SECONDS AFTER LEFT SYSTEM PRESSURE RECOVERS \geq 2,850 PSI
- ▶ IMMEDIATELY IF RIGHT SYSTEM PRESSURE DROPS $<$ 2,400 PSI

Hydraulic Shutoff Valves

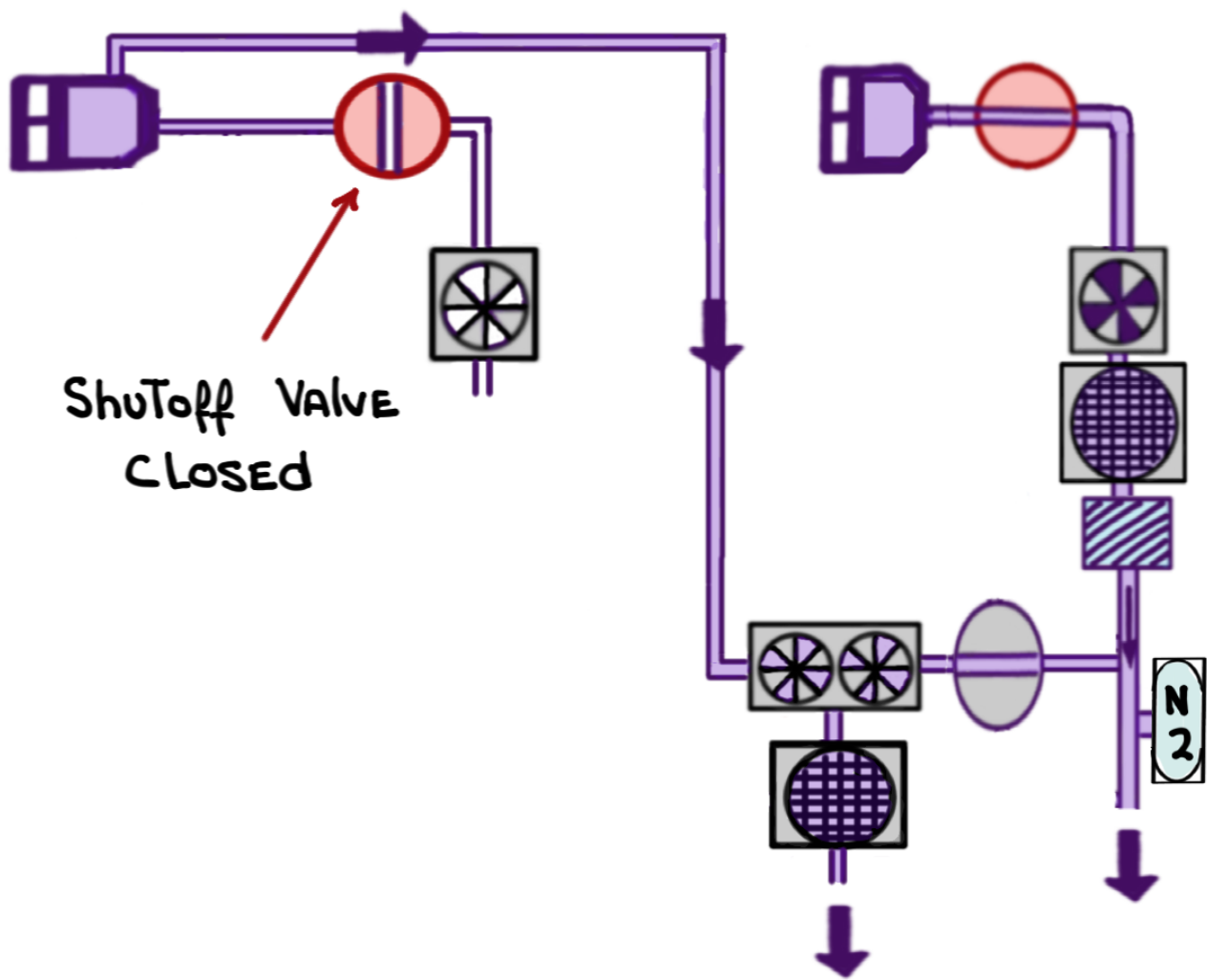
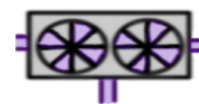
The hydraulic shutoff valves are located in the tail compartment and isolate the hydraulic fluid from the engine-driven pumps



The hydraulic shutoff valves are motor-operated and energized only when the **engine fire handles** in the cockpit are pulled

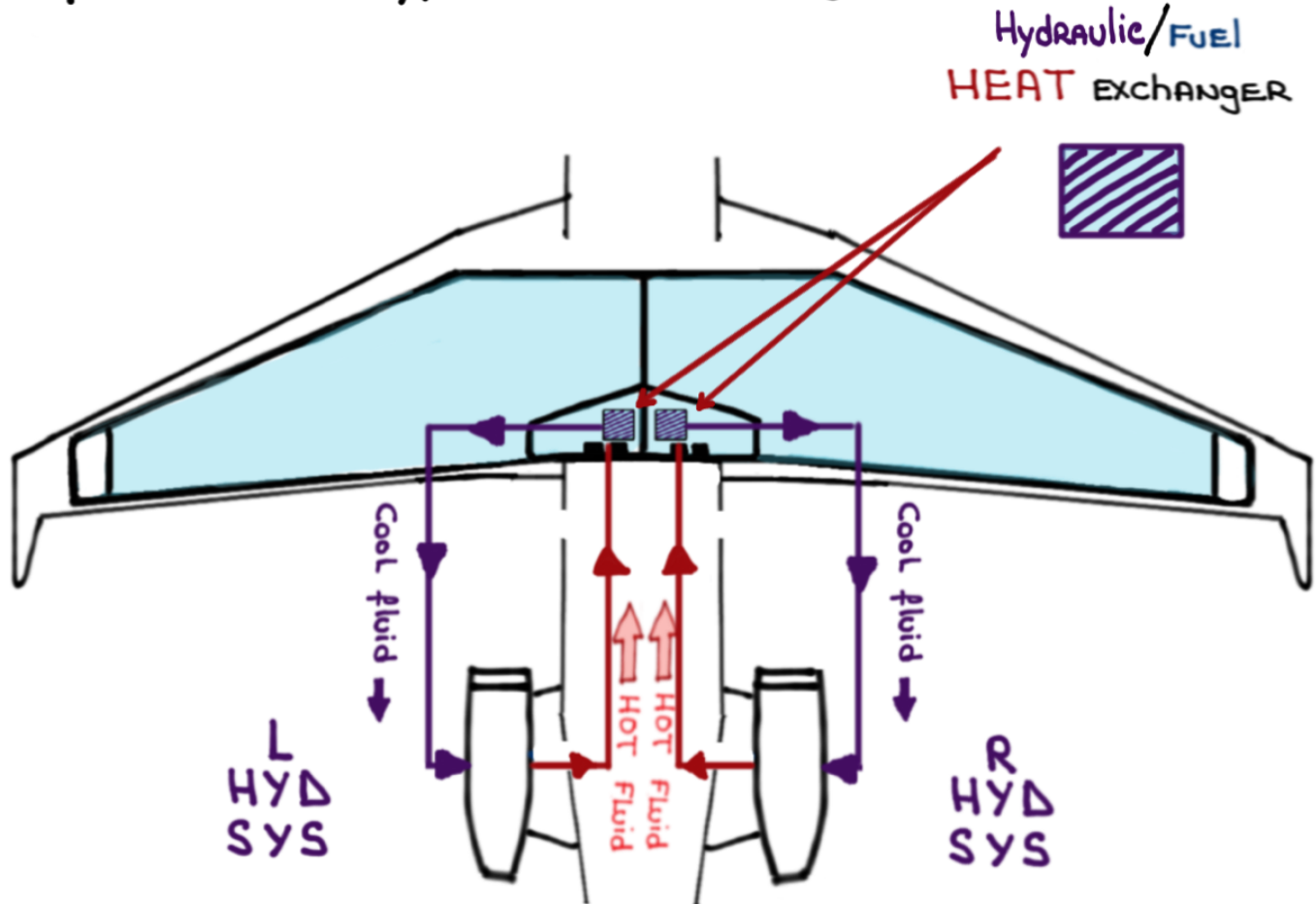


- Pulling The **LEFT FIRE HANDLE** does NOT shut off The supply of LEFT SYSTEM fluid To The **PTU**



Hydraulic fluid **HEAT** EXCHANGER

- THE HOPPER TANKS CONTAIN THE HYDRAULIC fluid - To - FUEL RADIATOR Type HEAT EXCHANGERS



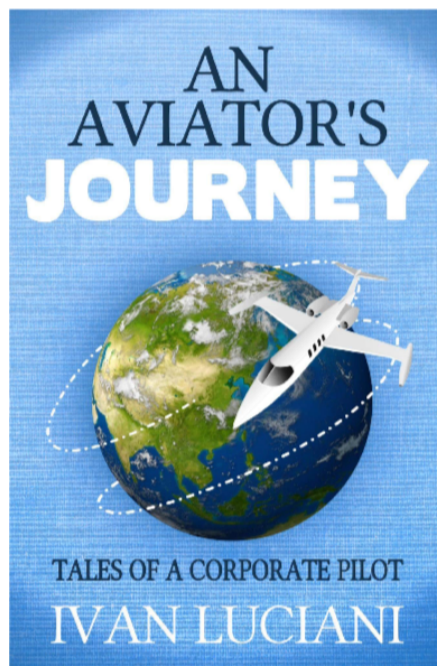
THE HEAT EXCHANGER UNIT IS INSIDE THE ONSIDE FUEL HOPPER. **HOT** hydraulic fluid flows continuously through the HEAT EXCHANGER without pilot input

HOT hydraulic fluid is cooled while **COLD** fuel in the HOPPER is warmed up

REMINDER: these system notes are intended for study purposes only. Always refer to official Gulfstream manuals and other approved references when operating your aircraft.

NOTE: these system notes are updated from time to time and what is posted on Code450.com will always be the most recent version.

Questions, comments or errors...please do send me an email:
ivan@code7700.com



Thank you!